

THE EFFECTIVE ALGORITHM FOR NAVIGATION OF QUADROTOR UAV IN INDOOR ENVIRONMENT

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Abstract: The quadrotor is an extensively researched and popular unmanned aerial vehicle in various fields, with navigation being one of the most widely studied areas. This paper focuses on setting the path for the quadrotor to move in a predefined 3D map environment. A real-time mathematical model for the quadrotor's motion was developed, incorporating the moment of resistance and the rotor rolling coefficient using improved Lagrange dynamic modeling. Two navigation scenarios were then built using Dijkstra and A star (A*) algorithms to evaluate the efficiency of the mathematical model based on criteria such as distance, time, and speed. The results indicate that Dijkstra's algorithm resulted in a 0.55% shorter travel distance for the quadrotor than the A* algorithm, while the quadrotor's flight time of the A* algorithm is 0.52% longer. However, the maximum speed of the quadrotor guided by the A* algorithm is 7.87% higher than that of Dijkstra's algorithm. These findings provide a basis for developing an optimal navigation strategy for the quadrotor to meet specific requirements.

KEYWORDS: Quadrotor UAV; navigation; dynamic modeling; Dijkstra algorithm; star algorithm.

1 Introduction

Robotics has emerged as a highly promising avenue for 4.0 industry [1-4]. The design of unmanned aerial vehicles (UAV) has rapidly evolved to meet the demands of various critical missions, including surveillance, fire protection, search and rescue, military, and civilian applications [5-7]. The quadrotor stands out as a versatile rotorcraft that can hover, fly forward, and take off and land vertically, making it an ideal research and application platform due to its adaptability, flexibility, and ease of construction [8, 9]. Many research groups have been focused on controlling and applying UAV in practical applications such as developing a navigation algorithm for quadrotor UAVs to guide themselves autonomously through unfamiliar tunnel-like environments [10], introducing an algorithm for autonomous navigation of a quadrotor UAV in a GPS-denied and unknown environment [11], using a vision-guided autonomous navigation method for quadrotor UAVs [12], introducing a robust geometric navigation algorithm for a quadrotor, based on the special Euclidean group, and formulated using the Newton-Euler method with ability of robustness against parameter uncertainty and disturbances, as well as convergence to the sliding manifold in a finite time and asymptotic convergence of the trajectory tracking [13]. A framework based on duelling double-deep recurrent Q-learning for achieving autonomous navigation in semi-known environments that is designed to handle situations where only a portion of the obstacles can be identified and located in advance [14]. It is suggested using two sensors to function as a flying odometer for estimating the forward velocity of a UAV [15]. The estimated velocity is then combined with INS using an Extended Kalman Filter to improve the precision of navigation. The authors suggested an improved navigation system for UAVs by integrating a monocular camera, an Ultra-Wideband (UWB) system, and INS [16]. They explore this integration in scenarios where UWB anchor density is low, with the aim of reducing the infrastructure cost associated with UWB system. Many methods have been employed for UAV path planning in indoor environments such as

UAV capabilities for indoor operations [17, 18], graph construction [19], rapidly exploring random trees [20], and Dijkstra's algorithm for optimizing UAV path [21]. Some groups have been developed simulation skill to evaluate the working ability for improving the service quality of robot [22] and autonomous vehicles [23]. This study deals with the creation of a path planning and obstacle avoidance system for a quadrotor UAV operating in an indoor environment. A Euler-Lagrange equation-based motion dynamics model for the quadrotor is developed in the study. The two navigation algorithms for quadrotor are used to assess the method's efficacy.

2 Materials and Methods

A quadrotor is an unmanned aerial vehicle that utilizes four rotors to generate lift. Its structure resembles a plus sign with two pairs of reversible rotors, where opposite pairs of rotors rotate in the same direction. In addition to contributing to the aircraft's thrust to gain altitude, these rotors also create motion around the center of gravity as shown in Figure 1. The quadrotor's movement is controlled by adjusting the speed of each rotor independently. By increasing the speed of two opposing rotors and decreasing the speed of two others, a moment is created that causes the quadrotor to tilt in a specific direction. This mechanism is utilized to control the quadrotor's motion in various directions.

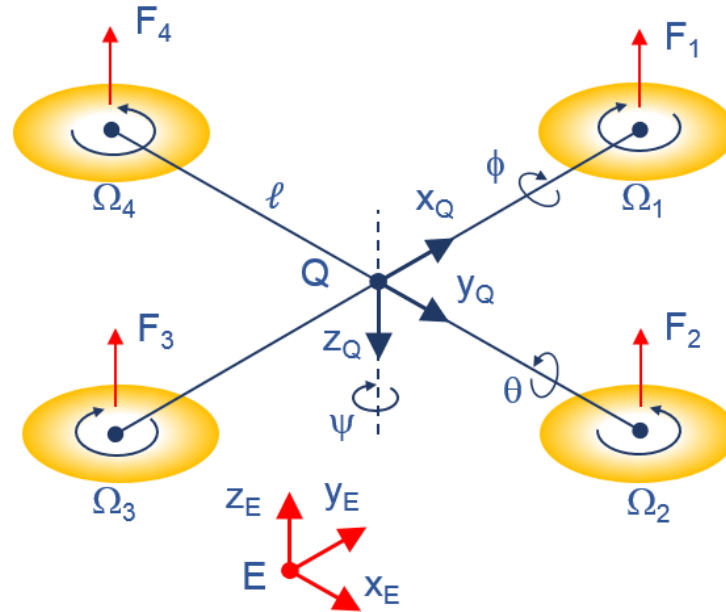


Fig. 1 The body-fixed frame and the inertial reference frame of quadrotor

Figure 1 illustrates the relationship between the inertial frame of reference, denoted as frame $\Psi(E^x_E y^y_E z^z_E)$, and the body-fixed frame, denoted as frame $\Upsilon(Q^x_Q y^y_Q z^z_Q)$. It also shows the distance between the rotor and the origin of the body-fixed frame, denoted as ℓ , and the lift forces, denoted as F_1, F_2, F_3, F_4 , produced by the rotors rotating at an angular velocity denoted as $\Omega_1, \Omega_2, \Omega_3, \Omega_4$.

The UAV quadrotor's movement in the world frame is expressed using Euler rotation sequence of Z-X-Y. This convention is used to construct the rotation matrix from frame Ψ to frame Υ :

$$\mathbf{R}_Y^\Psi = \begin{bmatrix} c_\psi c_\theta & c_\psi s_\theta s_\phi - s_\psi c_\phi & c_\psi s_\theta c_\phi + s_\psi s_\phi \\ s_\psi c_\theta & s_\psi s_\theta s_\phi + c_\psi c_\phi & s_\psi s_\theta c_\phi - c_\psi s_\phi \\ -s_\theta & c_\theta s_\phi & c_\theta c_\phi \end{bmatrix} \quad (1)$$

where $\begin{cases} c_* = \cos(*) \\ s_* = \sin(*) \end{cases}$, the angular velocity components of the robot in the body frame are denoted as $\omega_x, \omega_y, \omega_z$:

$$\omega_Y^\Psi = \omega_x \mathbf{x}_Q + \omega_y \mathbf{y}_Q + \omega_z \mathbf{z}_Q \quad (2)$$

Here, $\mathbf{x}_Q, \mathbf{y}_Q, \mathbf{z}_Q$ are unit vectors in the Y reference system, corresponding to the coordinate axes x_Q, y_Q, z_Q . The values of $\omega_x, \omega_y, \omega_z$ are obtained by differentiating the roll, pitch, and yaw angles using the following formula:

$$\begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix} = \begin{bmatrix} 1 & 0 & -s_\theta \\ 0 & c_\phi & c_\phi s_\phi \\ 0 & -s_\phi & c_\theta c_\phi \end{bmatrix} \begin{bmatrix} \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{bmatrix} \quad (3)$$

This section models the quadrotor's rotational dynamics using the Euler-Lagrange equation. The general dynamic model of the quadrotor is given by:

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\mathbf{q}}} \right) - \frac{\partial L}{\partial \mathbf{q}} = \tilde{\mathbf{Q}} \quad (4)$$

where $\mathbf{q}$ is the generalized coordinate, $\tilde{\mathbf{Q}}$ represents the generalized force, and Lagrangian is defined as L :

$$L(\mathbf{q}, \dot{\mathbf{q}}) = T - U \quad (5)$$

where $U = mgz$ is the potential energy (m is the mass of the quadrotor, g is the acceleration of gravity, z is the rotorcraft altitude), and the kinetic energy T is given by:

$$T = T_{rot} + T_{trans} = \frac{1}{2} \mathbf{\Omega}^T \mathbf{I} \mathbf{\Omega} + \frac{m}{2} \dot{\xi}^T \dot{\xi} \quad (6)$$

where $\mathbf{I} = \begin{bmatrix} I_{xx} & 0 & 0 \\ 0 & I_{yy} & 0 \\ 0 & 0 & I_{zz} \end{bmatrix}$ is the inertia matrix, $\dot{\xi}$ is the velocity matrix defined in the generalized coordinate:

$$\begin{cases} \mathbf{q} = (x, y, z, \phi, \theta, \psi)^T \in \mathbb{R}^6 \\ \xi = (x, y, z)^T \in \mathbb{R}^3 \\ \eta = (\phi, \theta, \psi)^T \in \mathbb{R}^3 \end{cases} \quad (7)$$

Angular velocity $\mathbf{\Omega}$ is given by:

$$\mathbf{\Omega} = \mathbf{W}_\eta \dot{\eta} \quad (8)$$

which yields $\mathbf{W}_\eta = \begin{bmatrix} 1 & 0 & -s_\theta \\ 0 & c_\phi & c_\phi s_\phi \\ 0 & -s_\phi & c_\theta c_\phi \end{bmatrix}$ and T_{rot} can be rewritten as follows:

$$T_{rot} = \frac{1}{2} \dot{\mathbf{q}}^T \mathbf{W}_\eta^T \mathbf{I} \mathbf{W}_\eta \dot{\mathbf{q}} = \frac{1}{2} \dot{\mathbf{q}}^T \mathbf{J}(\mathbf{q}) \dot{\mathbf{q}} \quad (9)$$

On the other hand, the generalized force $\tilde{\mathbf{Q}}$ is given by:

$$\tilde{\mathbf{Q}} = \begin{bmatrix} \mathbf{F}_\xi \\ \boldsymbol{\tau}_\eta \end{bmatrix} \quad (10)$$

$$\left\{ \begin{array}{l} \mathbf{F}_\xi = \mathbf{R}_Y^\Psi \hat{\mathbf{F}} = \mathbf{R}_Y^\Psi \begin{bmatrix} 0 \\ 0 \\ \sum_{i=1}^4 F_i \end{bmatrix} \\ \boldsymbol{\tau}_\eta = \begin{bmatrix} \tau_\phi \\ \tau_\theta \\ \tau_\psi \end{bmatrix} \square \begin{bmatrix} (F_3 - F_1)\ell \\ (F_2 - F_4)\ell \\ \sum_{i=1}^4 \tau_{M_i} \end{bmatrix} \end{array} \right.$$

Substituting (6) into (4) gives us the translational equation, and the rotational equation is given by:

$$\begin{cases} m\ddot{\boldsymbol{\xi}} + mg\mathbf{V}_z = \mathbf{F}_\xi \\ \mathbf{J}\ddot{\mathbf{q}} + \mathbf{C}(\mathbf{q}, \dot{\mathbf{q}})\dot{\mathbf{q}} = \boldsymbol{\tau}_\eta \end{cases} \quad (11)$$

where $\mathbf{V}_z$ is the rotor speed,

$$\mathbf{C}(\mathbf{q}, \dot{\mathbf{q}}) = \mathbf{J} - \frac{1}{2} \frac{\partial}{\partial \mathbf{q}} (\dot{\mathbf{q}}^T \mathbf{J})$$

The map is a 3D probability density function that serves as a scalar field, representing the probability density of occupancy at a specific location in space. However, the map is stored discretely as a representation of the most recent occupancy probability over time. In this paper, the construction of the grid map is discussed with the state model and the measurement model. A collision map is constructed, with a size determined during the initialization of the map. A 3D collision map is created with a specified size, describing the density of the occupied parameters on the x, y, and z axes as shown in Figure 2. The list of squares is then looped through, and for each block, its corresponding position on the map and state are calculated by transforming coordinates between real and design space. The map has been arranged according to the layout depicted in Figure 2, allowing for the observation of cell occupancy density. The colors have been incorporated for ease of visual examination.

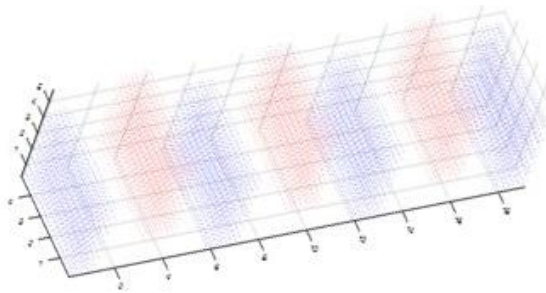


Fig. 2 Building 3D maps

The two widely used algorithms are the A* algorithm and Dijkstra algorithm. The paper involves a combination of these two algorithms to determine the optimal direction for quadrotor navigation. The A* algorithm constructs a path tree from the initial node and progressively expands paths until one leads to the target node. During each iteration of the algorithm, A* selects neighboring nodes with the lowest cost using an evaluation function.

Algorithm 1: A* algorithm

```

1.    $A^*(Start, Goal)$ 
2.    $S \leftarrow \emptyset$ 
3.    $Q \leftarrow V$ 
4.    $G(Start) \leftarrow 0, H(Start) \leftarrow H_c(Start, Goal),$ 
       $F(Start) \leftarrow H(Start)$ 
5.   while  $Q \neq \emptyset$  do
6.      $C\_node \leftarrow EXTRACT - MIN - F(Q)$ 
7.     if  $C\_node = Goal$  then
8.       return BestPath
9.     for each  $N \in C\_node$  do
10.      if  $N \in S$  then Nothing
11.      else if Calculate  $N_s, G, H, F$ 
12.        if  $G(N \in Q) > calculated\ G(N)$  then
13.           $RELAX(N, N \in Q, w)$ 
14.           $N_s\_parent = C\_node$ 
15.          Add  $N$  to  $Q$ 
16.        else then  $calculated\ N_s, G, H, F$ 
17.           $N_s\_parent = C\_node$ 
18.          Add  $N$  to  $Q$ 
19.        end if
20.      end if
21.    end for
22.  end if
23. end while

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Algorithm 2. Dijkstra algorithm

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1.    $DIJKSTRA(V, E, w, s)$ 
2.    $S \leftarrow \emptyset$ 
3.    $Q \leftarrow V$ 
5.   while  $Q \neq \emptyset$  do
6.      $u \leftarrow EXTRACT - MIN(Q)$ 
7.      $S \leftarrow S \cup \{u\}$ 
9.     for each  $v \in Adj(u)$  do
10.       $RELAX(u, v, w)$ 
13.    end for
14.  end while

```

Dijkstra's algorithm can be used to find the shortest path in a weighted, directed graph with non-negative weights. The algorithm maintains a priority queue called $MIN(Q)$, which stores the unprocessed vertices along with their shortest path estimates. It repeatedly extracts the vertex with the minimum value from $MIN(Q)$ and relaxes all edges that are incident from u to any vertex in $MIN(Q)$. Once a vertex is extracted from $MIN(Q)$ and all its incident edges are relaxed, the algorithm considers the vertex as processed and does not touch it again. The algorithm stops when $MIN(Q)$ is empty or when every vertex is examined exactly once.

3 RESULTS AND DISCUSIONS

This section establishes a flight map for the quadrotor, which includes dimensions and obstacles depicted in Figure 3.

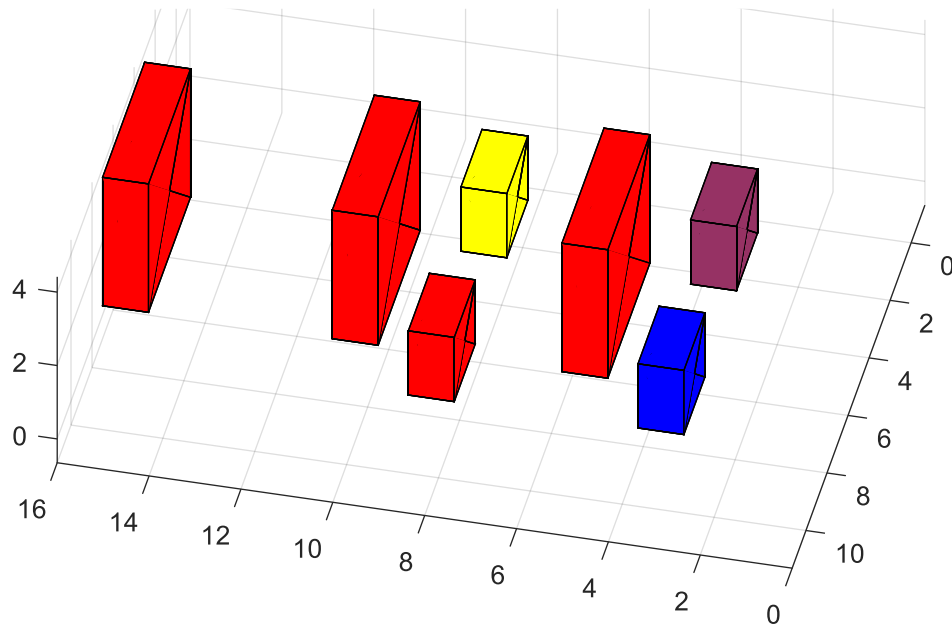


Fig. 3 Quadrotor's movement map

The quadrotor UAV travels from its initial position at coordinates (0, 6, 0) to the destination point at (14, 4, 2). The robot's mass is $m = 0.03kg$, $\ell = 0.046m$, and the maximum tilt angle is 40° . The A* algorithm is employed to determine the actual position, velocity, and path of the quadrotor, which are represented in Figures 4-7. The Dijkstra's algorithm is utilized to illustrate the actual position, velocity, and path of the quadrotor as shown in Figures 8-11.

The results are obtained using a sampling time of 0.2 seconds during UAV flight. The quadrotor takes 9.60s to travel from the starting point to the destination using the A* algorithm, while it takes 9.55s by using Dijkstra's algorithm. Therefore, Dijkstra's algorithm results in a 0.52% faster travel time than the A* algorithm. In Figures 7 and 17, the blue line represents the actual path of quadrotor. The A* algorithm's path is 3.5% shorter than Dijkstra's algorithm path. However, according to Figure 6 and Figure 10, the quadrotor travels a distance of 18.03m with the A* algorithm and 17.93m for Dijkstra's algorithm, resulting in Dijkstra's algorithm being 0.55% shorter than the A* algorithm in terms of travel distance. However, with Dijkstra's algorithm, the quadrotor's movement is unstable due to deceleration to ensure the correct target, as shown by the path. Figures 5 and 9 illustrate that the quadrotor achieves a maximum speed of Dijkstra's algorithm of 7.87% higher than that of the A* algorithm.

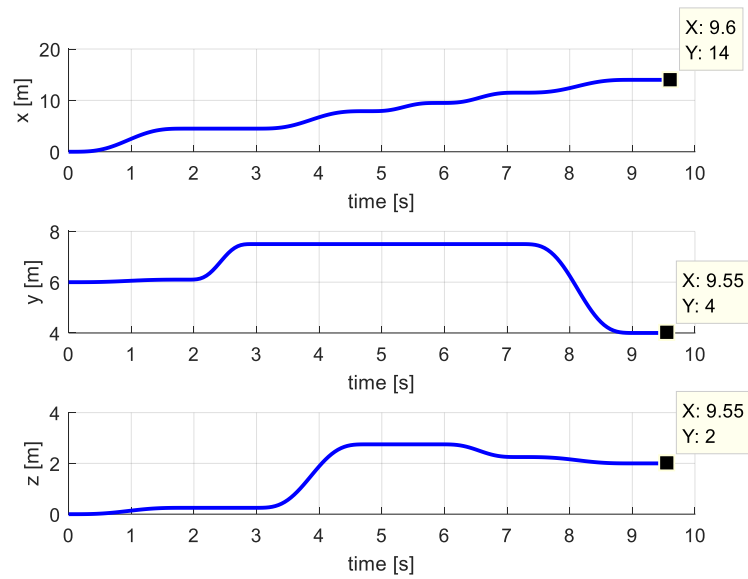


Fig. 4 Position of robot during movement using A* algorithm

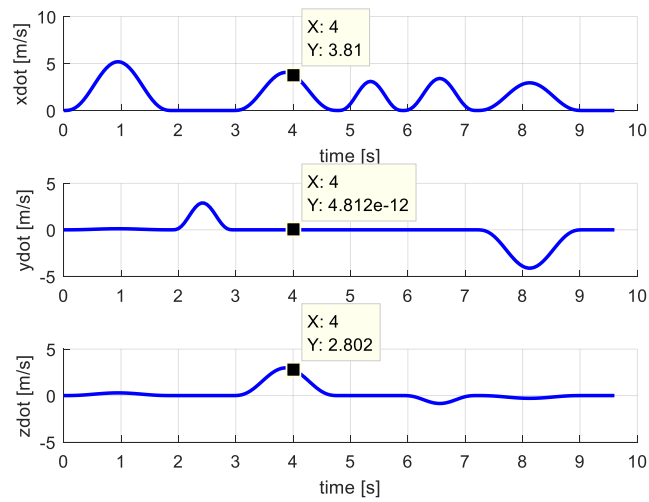


Fig. 5 Velocity of robot during movement using A* algorithm

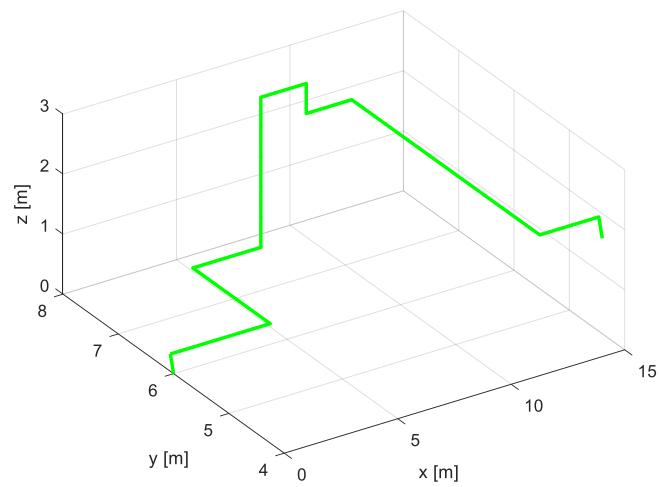
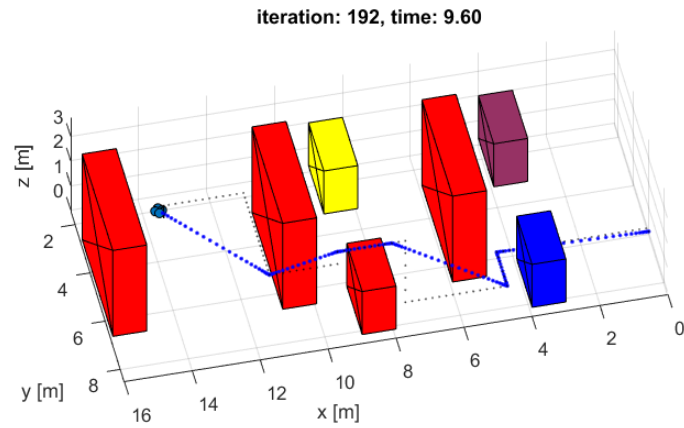
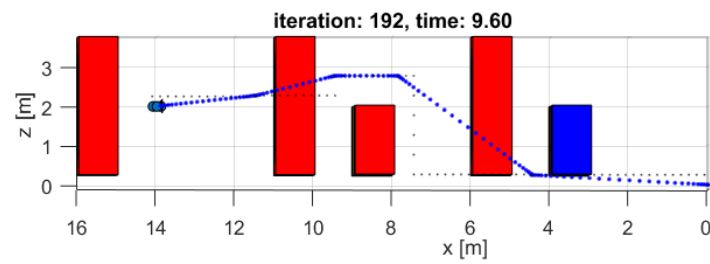


Fig. 6 Path of the quadrotor UAV using A* algorithm



(a)



(b)

Fig. 7 Actual path of quadrotor when using A* algorithm:
a- Panoramic view; b- view angle z-x

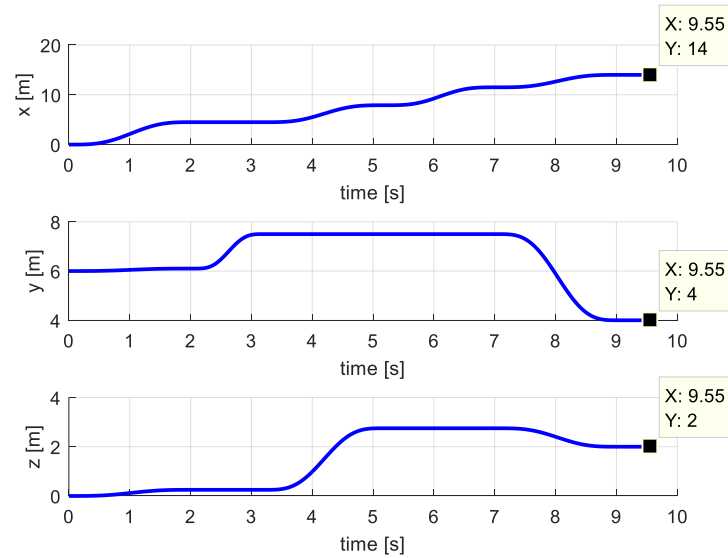


Fig. 8 Position of robot during movement using Dijkstra's algorithm

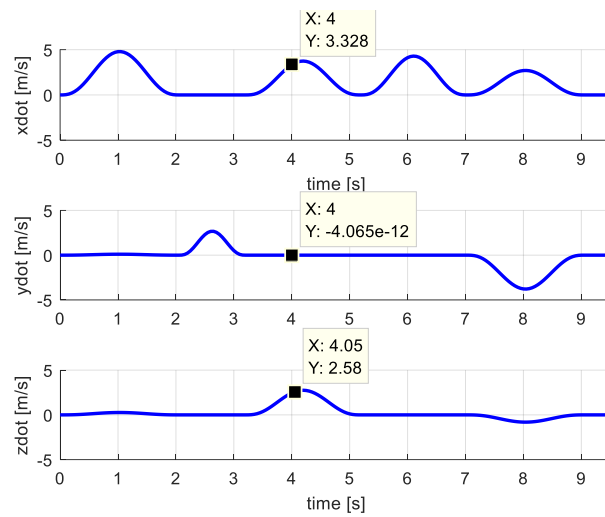


Fig. 9 Velocity of robot during movement using Dijkstra's algorithm

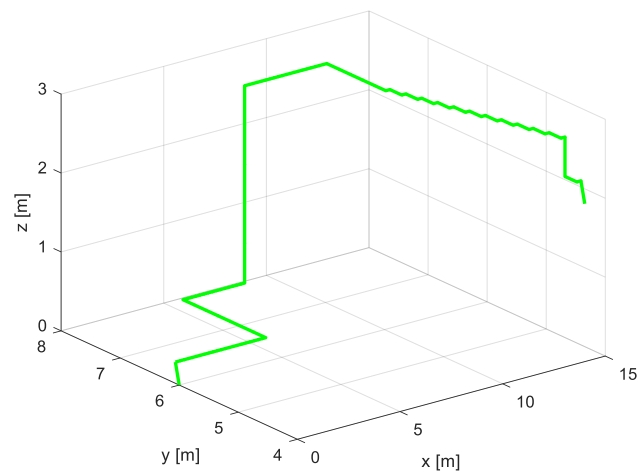
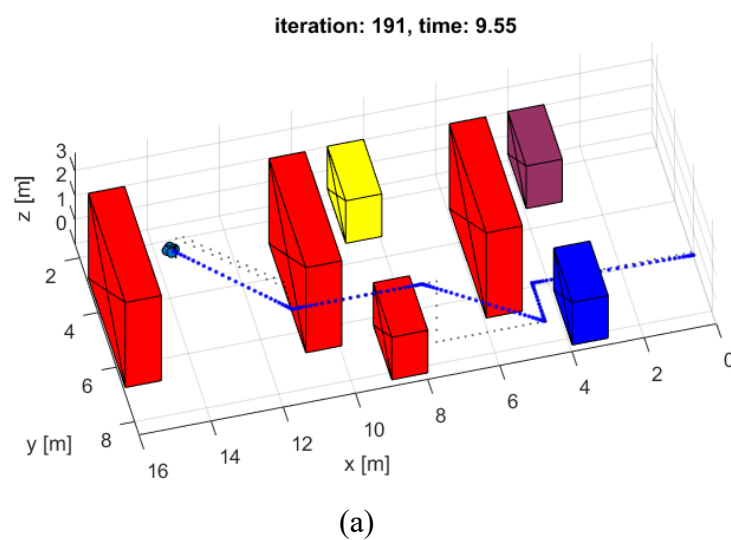
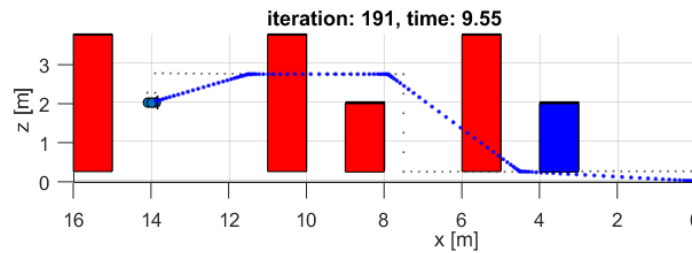


Fig. 10 Path of the quadrotor UAV using Dijkstra's algorithm





(b)

Fig. 11 Actual path of quadrotor when using Dijkstra's algorithm:
a- Panoramic view; b- view angle z-x

CONCLUSIONS

This study investigated the performance of A* and Dijkstra algorithms in determining the path and travel time of a quadrotor UAV from a starting point to a destination. The results show that while A* algorithm resulted in a path that is 3.5% shorter than that of Dijkstra's algorithm, Dijkstra's algorithm covered a travel distance that is 0.55% shorter than A* algorithm. However, using Dijkstra's algorithm resulted in unstable movement due to deceleration to ensure the correct target. The quadrotor achieved a maximum speed that is 7.87% higher when using A* algorithm compared to when using Dijkstra's algorithm. For flight time, Dijkstra's algorithm resulted in a 0.52% faster travel time than A* algorithm. Findings suggest that the choice of algorithm depends on the specific objectives and constraints of UAV flight mission.

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